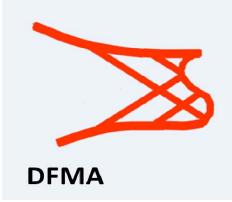
Digital Fabric Mechanics Analyzer



Youqi Wang Department of Mechanical &Nuclear Engineering Kansas State University Manhattan, KS 66506



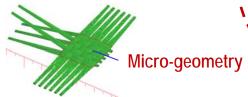
Applications

K-State

Composites

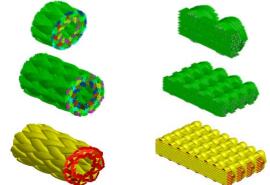
Micro-geometry & Fiber damage

Textile process simulation
 Static Simulation (Weaving)



Dynamic relaxation with periodic boundary conditions (DFMA: Fabric mechanics.com)
 Unit cell fiber-level micro-geometry

- − Assembly of unit cells → Fabrics
- Yarn Geometry



Dynamic Simulation (Weaving+ Beat-up)

- Fabric stress analysis
 - Static Analysis: fiber stress, fabric deformation, fabric strength, fabric damage
 - Dynamic Analysis
 Impact

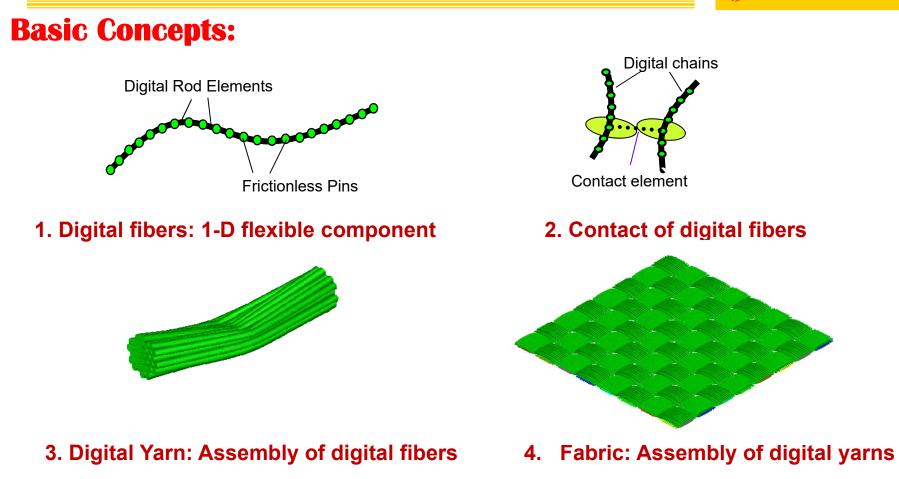






Fiber-level fabric Mechanics Analysis

K-State Composites Laboratory

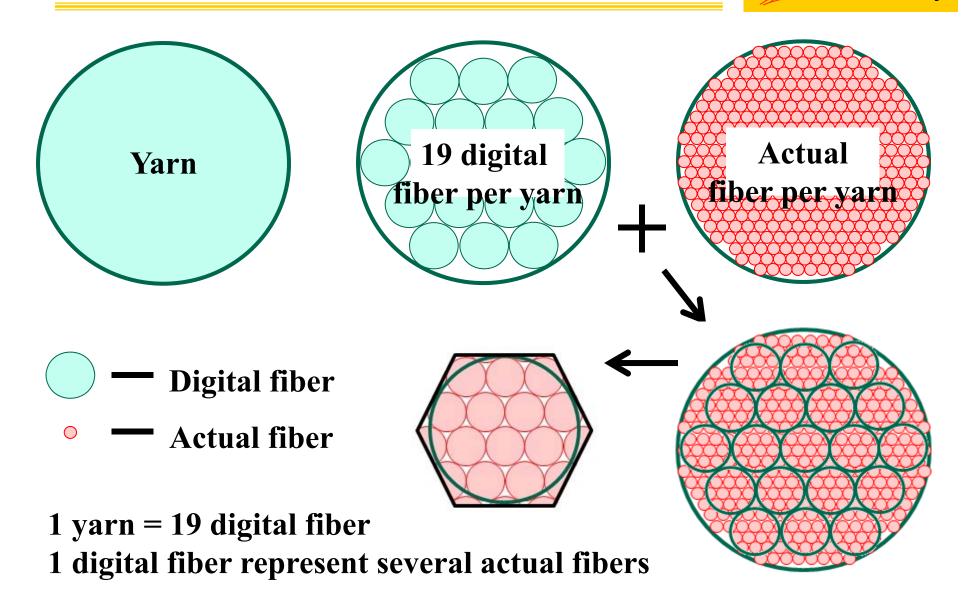


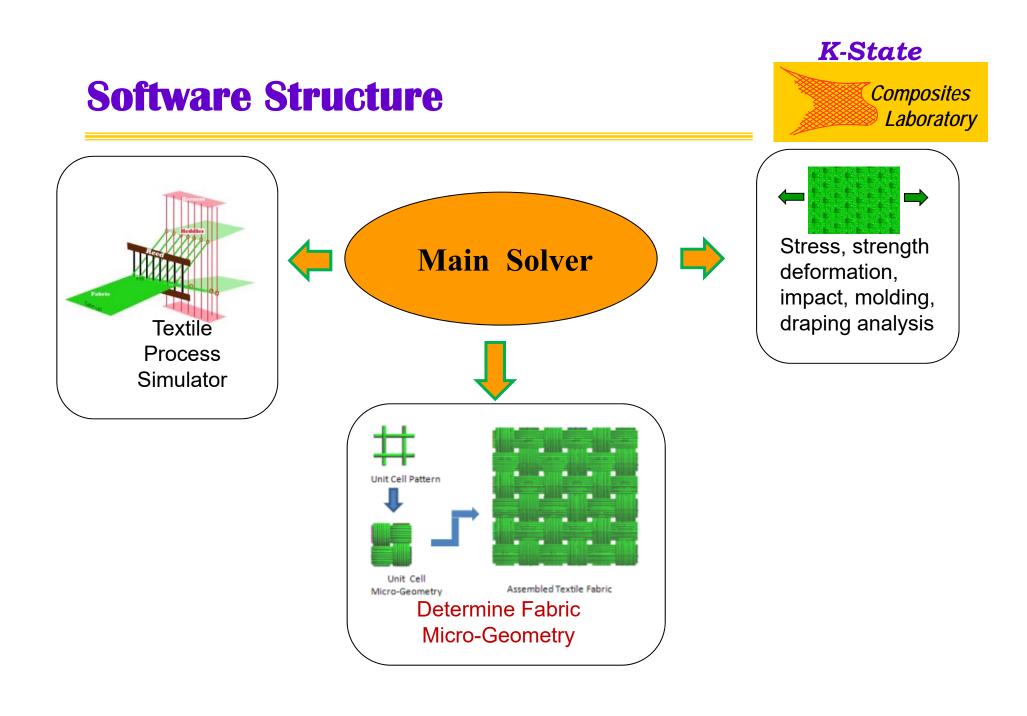
Assumption on Digital Yarns: Length of all digital fibers inside a digital yarn are the same. (Iso-length assumption)

Digital fibers and Actual fibers

Composites

K-State





Part 1: Interface for Fabric Micro-Geometry

le View Plot Window Help										
■ Carla Car		Unsmoothen oss Section		井 Line 📕 Fiber D-Fiber Model	🗱 Surface Mesh <mark>—</mark> Yarn Surface Solid Yarn Model	e 🛞 🚥 Remesh	ting the second	😰 🖟 Pick Yarn	2 Tutorial	
Fabric Micro-geometry Define Yarn Properties -Yarn Type Properties Unit Cell Topology -2-D Woven -3-D Woven -3-D Braided Digital Element Mesh And Remesh -Fibering Elementing -Scaling Periodical Geometry Relaxation -Node Images Boundary Conditions -Tension Properties Relaxation Parameters Solution -Start (Restart) -Continue Rework Solid Yarn Model -Yarn Model Parameters Generate Yarn Model Display Yarn/Yarn Type Display Section Display Recover Whole Geometry Fabric Assembly Single-Layer Assembly Based on Unitcell Number Based on Fabric Size Shape and Orientation Multi-Layer System Recover the Unit Cell	2									
	×		Fiber Vol	:: 0.002505 m ume Fraction: 0.49409	7 : fabric/cell is: 31562828.783618 g/n					

Unit Cell Micro-Geometry



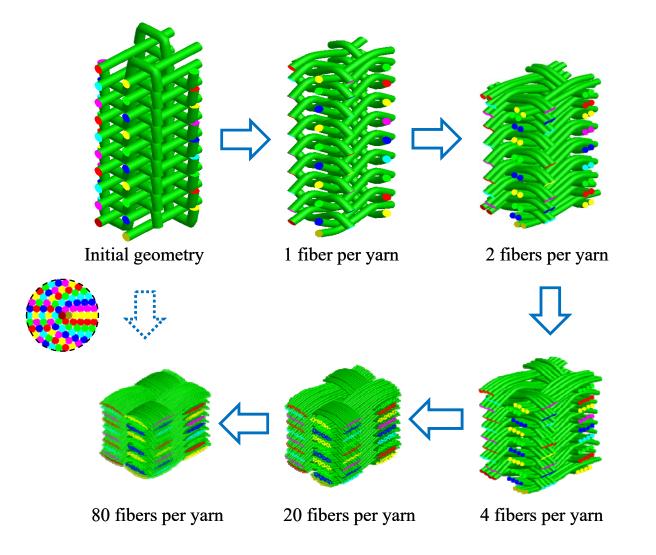
Laboratory

Cell Topology with Unit cell **Cell Topology Tow Structure** assigned yarn structure micro-geometry Plain ╈ Twist ╋ Multi-ply twist ٦. +

Micro-geometry is determined by dynamic relaxation approach with periodic boundary conditions.

3-D Woven Fabric Unit Cell Micro-geometry

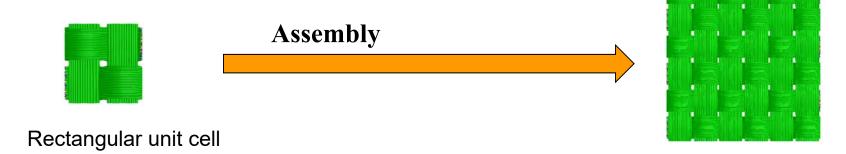
K-State





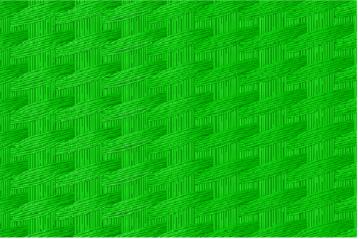
K-State

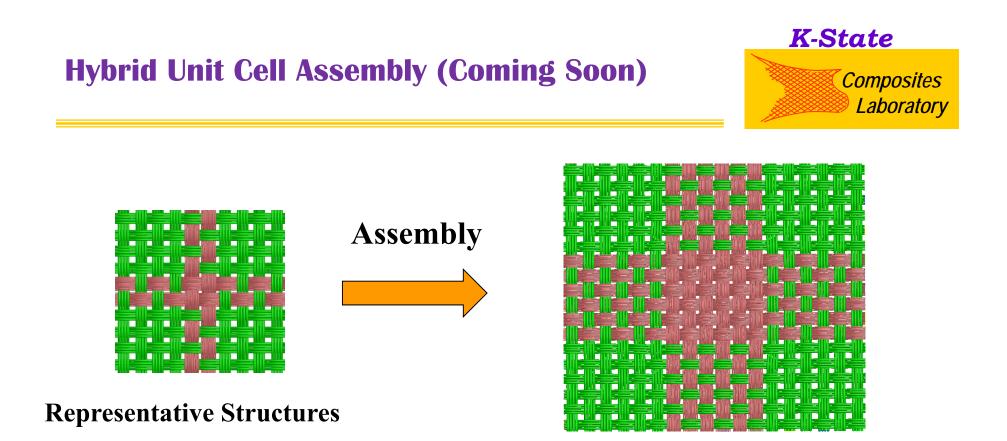
Composites





Non - rectangular unit cell (Angle-interlock)

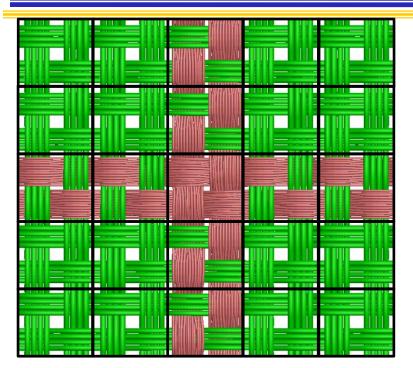




Expanded Fabrics

The representative structure of a fabric is not the unit cell of the fabric. It contains all cells required to assemble a hybrid fabric.

Preprocessor 1: Basic cells in hybrid meshes



5, 1	5, 2	5, 3	5, 4	5, 5
4, 1	4, 2	4, 3	4, 4	4, 5
3, 1	3, 2	3, 3	3, 4	3, 5
2, 1	2,2	2, 3	2, 4	2, 5
1, 1	1, 2	1, 3	1, 4	1, 5

K-State

Composites

Laboratory

A two-level hybrid mesh include 25 basic cells:

- 1. Yellow cells: with periodic boundary in x- and y-directions
- 2. Light blue cells:
 - with periodic boundary conditions in y-direction
 - Ieft and right boundary displacements: defined by the neighboring cells
- 3. Deep blue cells:
 - with periodic boundary conditions in x-direction
 - top and bottom boundary displacements: defined by neighboring cells
- 4. White cells:
 - all boundaries: defined by neighboring cells

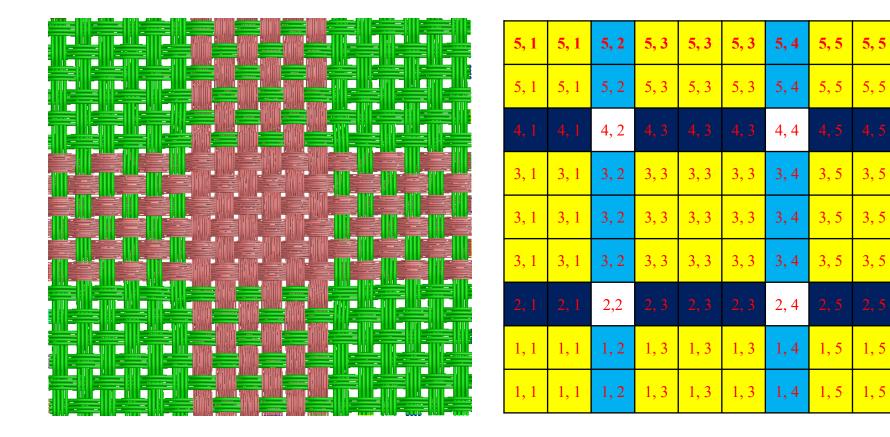
Relaxation sequence: 1. yellow cells, 2. blue cells, 3. white cells

Preprocessor 2: Fabric Assembly

K-State

Composites Laboratory

Hybrid Mesh Assembly Map:



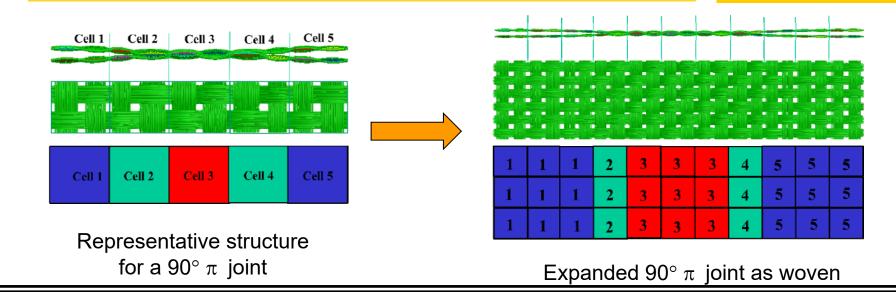
Fabric with 81 cells

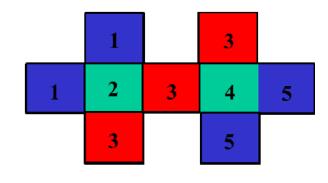
Assembly Map

K-State

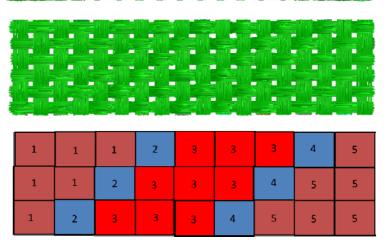
Hybrid Assembly of π Joint (Coming soon)

Composites





Representative structure for a $45^{\circ} \pi$ joint

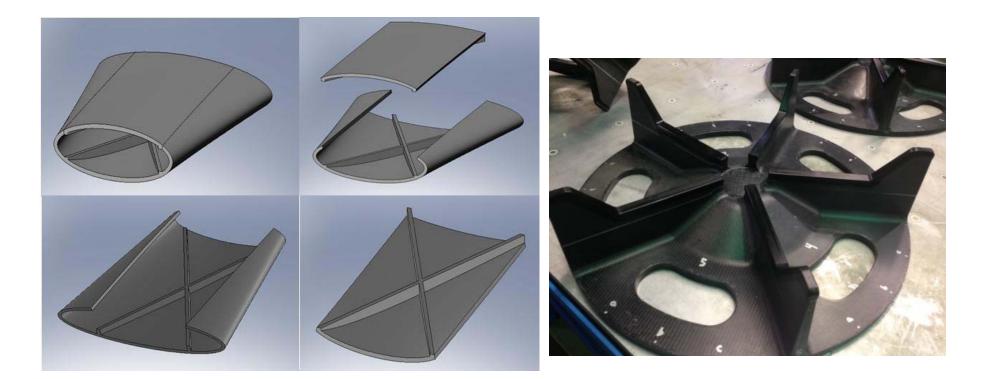


Expanded 45° π joint as woven

Goal of Hybrid Assembly



• Generate complex structure such as:



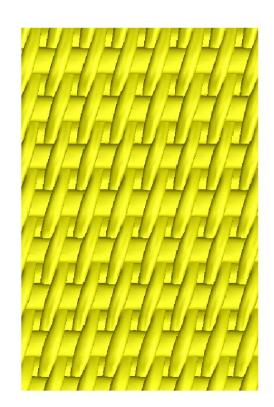
Fabric Surface comparison

K-State



Microscopic picture

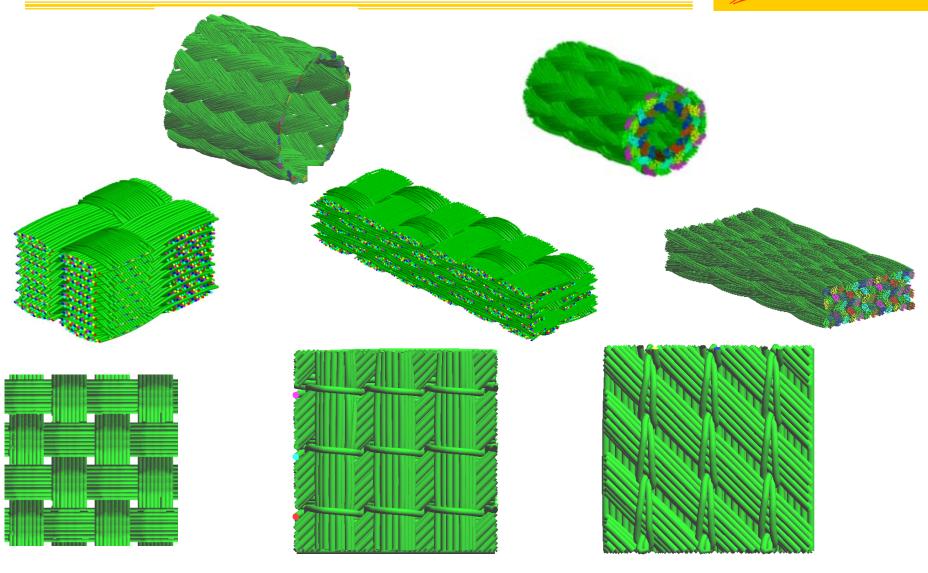
Assembly with fiber mode

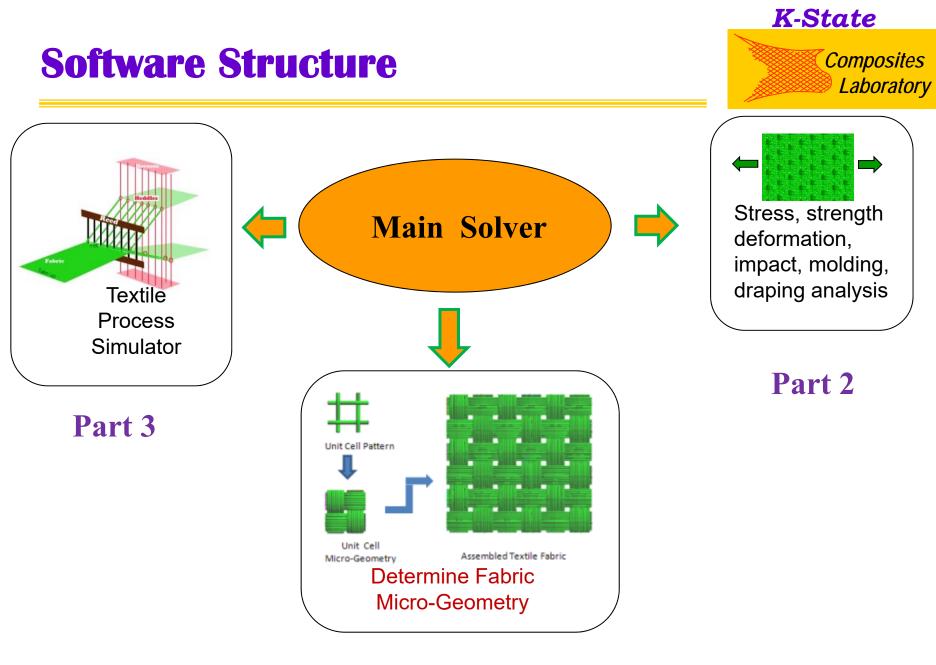


Assembly with solid yarn mode

Various Fabric Micro-Geometries

K-State





Part 1

Part 2: Stress, Deformation, Impact Analysis

DEA Fabric Mechanics Analyzer - Fabric Stress Ana	alysis					
ile View Plot Window Help	\bigcirc Smoothen \bigcirc Unsmoothen $ $ \approx S Fiber Cross Section	moothen 🗠 Unsmoothen Fiber Path	 Display Projectile	🛞 🚥 Remesh	solve Pick Yarn	? Tutorial
 Fabric Stress Analysis Yam Properties Layer Type Friction Coefficient Table Load and Initial Conditions On Node On Boundary Apply Constraint Apply Colocity Apply Coulping Only On Line or Curve (Coming soon) On Area (Coming soon) On Yam Type On Yam Type On Yam Rigid Bodies Spherical Particles Circular Cylinders Plate Solid Bodies Any shape Dynamic Analysis Start Continue Rework Display Contour Stress Contour Stress Range Return to Fabric Plot 						

Fabric Stress, Deformation and Strength Analysis

The software has been used for the following area:

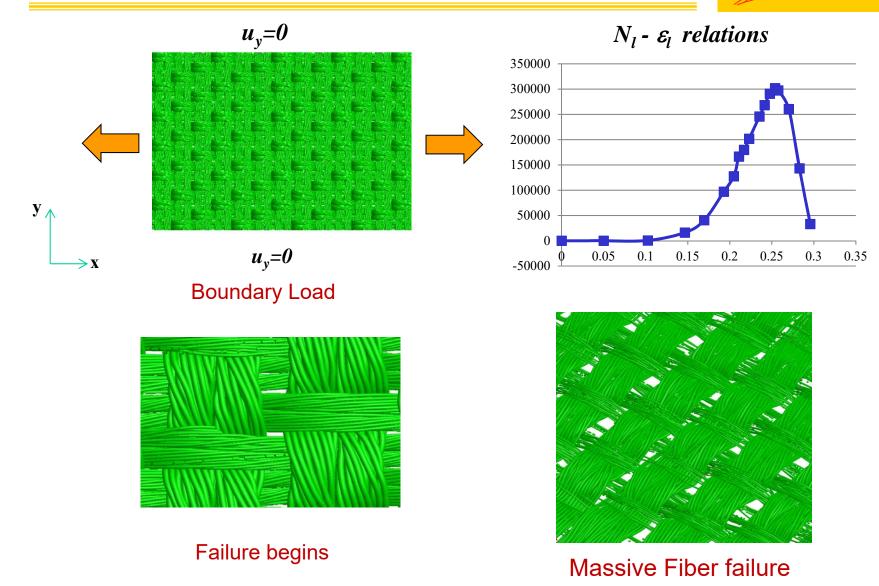
- 1. Stress and strength analysis of fabrics
- 2. Draping process
- 3. Molding process:
 - Helmet manufacturing process
 External mold and internal plug
 - Airplane Engine Blade
 - Double moulding and net-shape forming
- 4. Ballistic penetration process
 - PC based simulation
 - Cluster based simulation (Parallel code)
 - Projectiles: Particles, rigid body, and elastic body
- 5. Fiber properties:
 - Elastic or elasto-plastic in both longitudinal and transverse direction
- 6. Post process and Virtualization Capability:

Displacement and stress

K-State

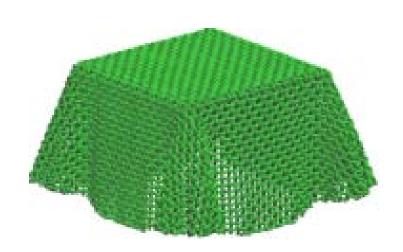
Example 1: Fabric Stress and Strength Analysis

K-State



Example 2: Draping Process

K-State

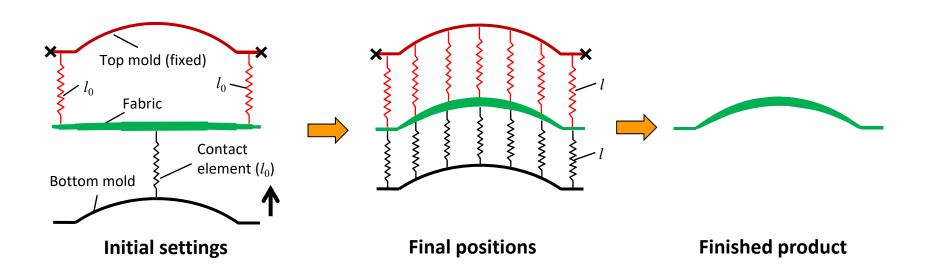




Net Shape Composite and the Molding Process —Dynamic molding process

K-State

Composites Laboratory



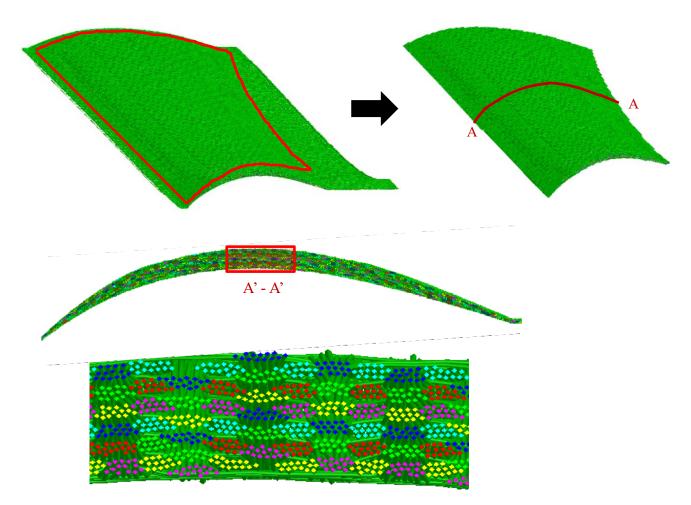
- Top mold fixed, bottom mold free to move
- Shortest distance between fabric and both molds is l_0
- l_0 is defined contact element length
- All contact elements are equal
- Remove both molds and relative contact elements

4. Net Shape Composite and the Molding Process



Laboratory

Airplane engine blade



Example 3: Molding Process (Helmet)

K-State

Composites

Upper mold: Made of metal materials

Lower mold: Plug (Rubber like material)

Simulation process:

- Drape the fabric on the lower mold(plug)
- The lower mold move upward until the pressure between upper mold and the lower mold reach a certain value.

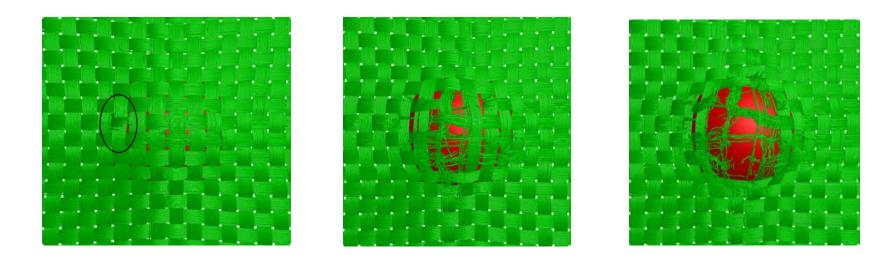


Helmet

Ballistic Penetration Simulation



Composites



Failure Starts

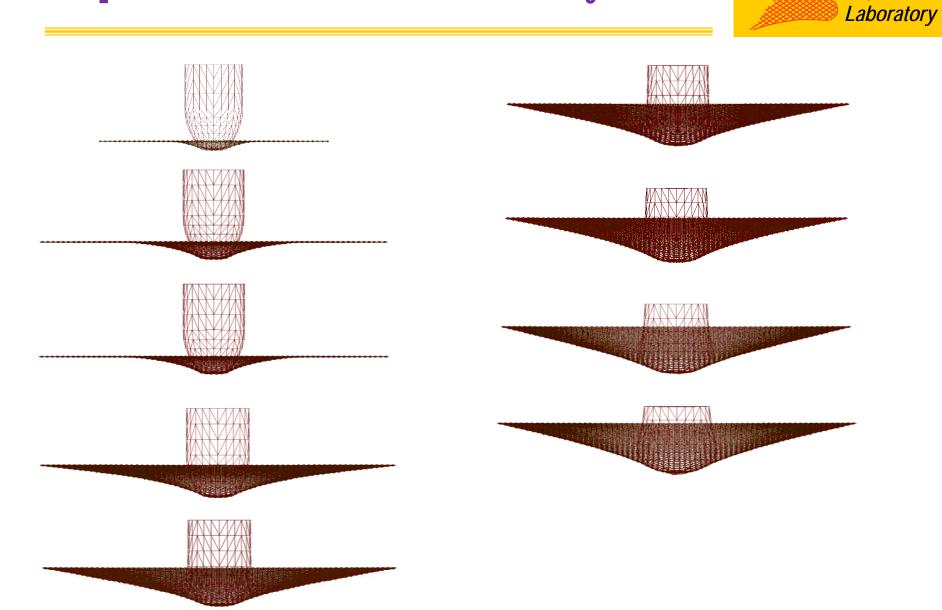
Progressive Failure

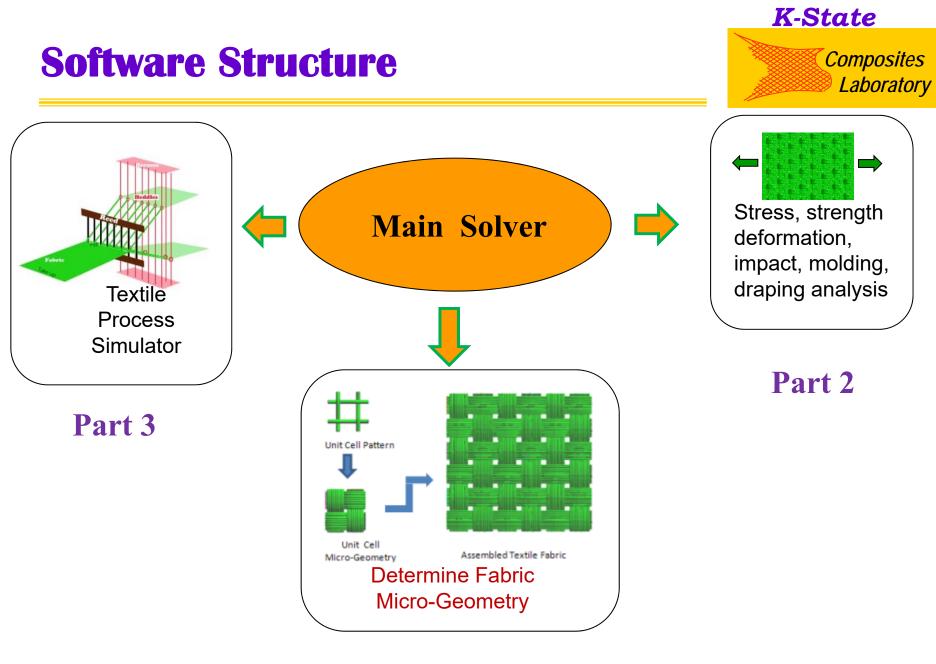
Prior to Perforation

(PC Based and Cluster Based)

Impact with Deformable Body

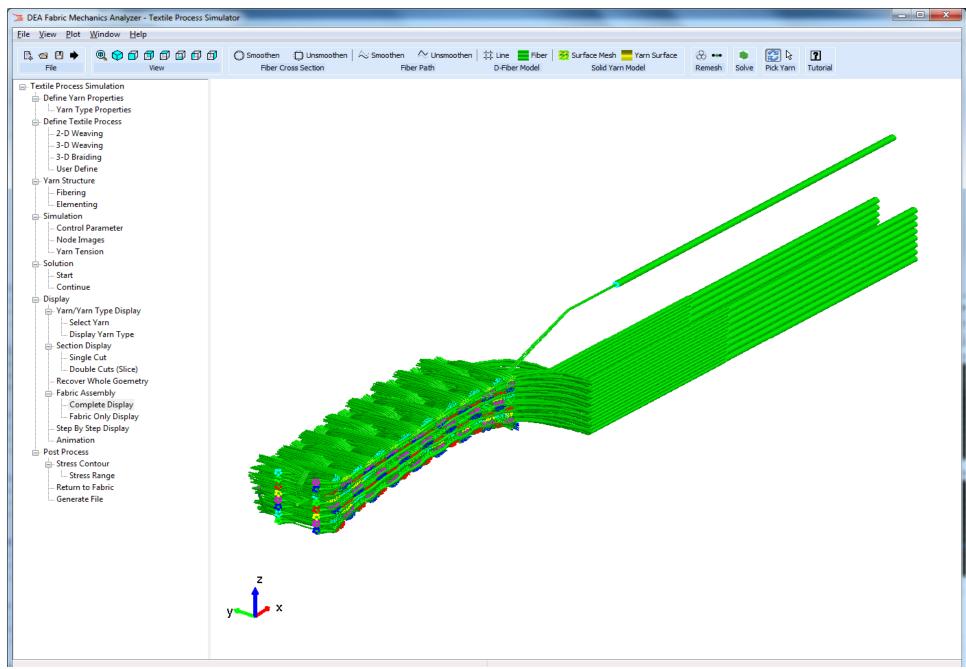
K-State Composites





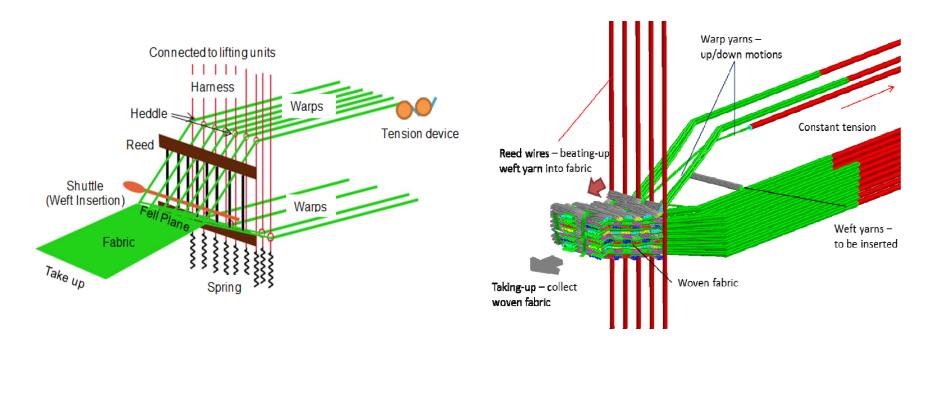
Part 1

Part 3: Weaving Process Simulator



3D WEAVING MACHINE STRUCTURE AND WEAVING PROCESS SIMULATION

K-State





Applications of Process Simulation



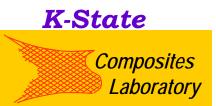
- 1. Simulate weaving process. Analyze weaving kinetics, such as yarn tension, beat-up velocity, beat-up force, take up frequency, interfiber friction, reed-fiber friction, on fabric thickness, unit cell length and unit-cell width.
- 2. Analyze fiber damage during the weaving process. The beat-up velocity and reed-fiber friction can play important role on the fiber damage. Fiber damage could affect the fabric strength and composite strength.
- 3. We have collaborated with 3-D weaving company to design 3-D weaving machine for 3-D armors. The simulator is used to calculate the beat-up load applied to the 3-D weaving machine. It will guide company for the design of new machine.
- 4. It can also be used to simulate braiding processes.

Software Version



- Window based version(Free download):
 - Have a user friendly graphic interface
 - Have pre-processor and post processor
 - Used open-mp (parallel code with shared memory) and C++ language.
- Cluster based version:
 - Use MPI and C++ language
 - Can be used for large scale simulation
 - Have to use the window-based version for pre- and postprocessors

Future development of DFMA



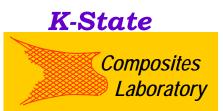
Our research interests:

1. Textile composite analysis:

Feasible digital fiber level micro-mechanics model for textile composites

- 2. Problem in manufacturing process:
 - Resin flow in textile composite preform
 - Gas infiltration in textile composite preform
 - Vapor deposition process





Financial supports from US Army Research Laboratory under contracts W911NF-05-2-0010, W911NF -08-2-0041 and W911NF-12-2-0020 are acknowledged.

Thank You!

